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Buckling, Deflection, And Twist By M. A. Ferman**

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1 of 11 American Institute of Aeronautics and Astronautics Supersonic Wing Design Method Using an Inverse Problem for Practical Application Shoji Sakashita1, Takumi

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Abstract. An inverse method for supersonic wing design has been developed. The present method is based on iterative residual-correction concept.

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An Inverse Design Method for Wings Using Integral Equations and Its Recent Progress Kisa Matsushima* and Susumu Takanashi** (f), * FUJITSU Ltd., 1-9-3, Nakase, Mihama

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Patent wo1996003215a2 - lift enhancing tip vortex

Embodiments include a twist (66) along the span of a foil (82b), and anhedral tip droop having a divergent axis of droop curvatur (86), a movable vortex flap

The development of a multi-disciplinary wing

The Advanced Aerodynamics Department at Bombardier Aerospace is developing a multi-disciplinary methodology for the design of commercial transport wings.

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Supersonic inverse design method for wing-fuselage

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Patent us5634613 - tip vortex generation

Methods for using tip generated vortices to improve performance of foils. These methods include generating a substantially streamwise beneficial vortex (74) near the

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